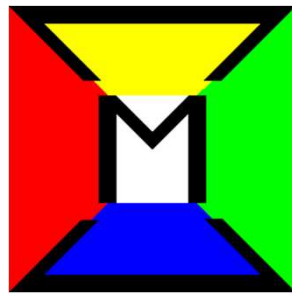


# AASHTO Strength Design Checks



*[www.mbrace3d.com](http://www.mbrace3d.com)*

## Purpose

1. Understand the different AASHTO strength design checks (10<sup>th</sup> Edition, 2024).
2. Describe how mBrace3D (v4.0) performs these checks automatically.
3. Illustrate these new capabilities on AISC/NSBA Design Example 4, a typical curved, continuous, plate girder bridge.

## Method Outline

mBrace3D v4.0 enables the automation of a variety of AASHTO LRFD strength design checks, by:

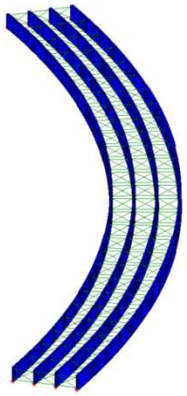
I. Extracting moment and shear results from all the required load cases:

<u>DCI-S</u>	Effects due to the self-weight of the steel superstructure only
<u>DCI-C</u>	Effects due to the wet concrete load (before composite action develops)
<u>DC2</u>	Effects due to superimposed dead loads (parapets, barriers)
<u>DW</u>	Effects due to the wearing surface load
<u>LL+IM</u>	Live load effects, including dynamic allowance

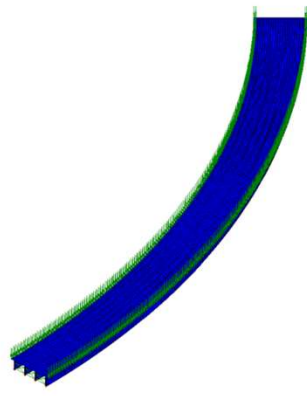
2. Computing the corresponding principal bending ( $f_b$ ) and lateral bending ( $f_l$ ) stresses, using the appropriate section moduli (non-composite section, short-term and long-term composite sections)

3. Evaluating the strength design checks from the AASHTO LRFD Bridge Design Specifications (10<sup>th</sup> Edition)

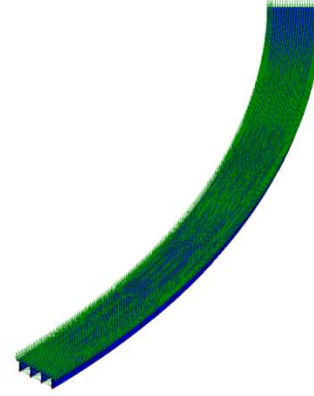
## Step I – Load Cases



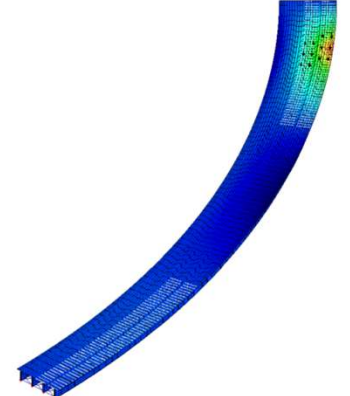
Steel superstructure only  
DCI-S, DCI-C



Long-term composite section  
DC2



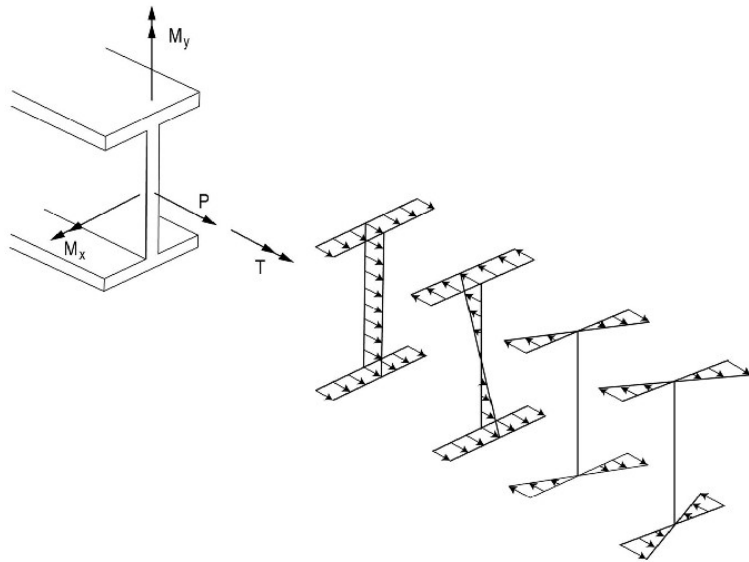
Long-term composite section  
DW



Short-term composite section  
LL+IM

For LL+IM, the envelope moment and shear values are derived using an influence surface analysis.

## Step 2 – Principal and Lateral Bending Stresses



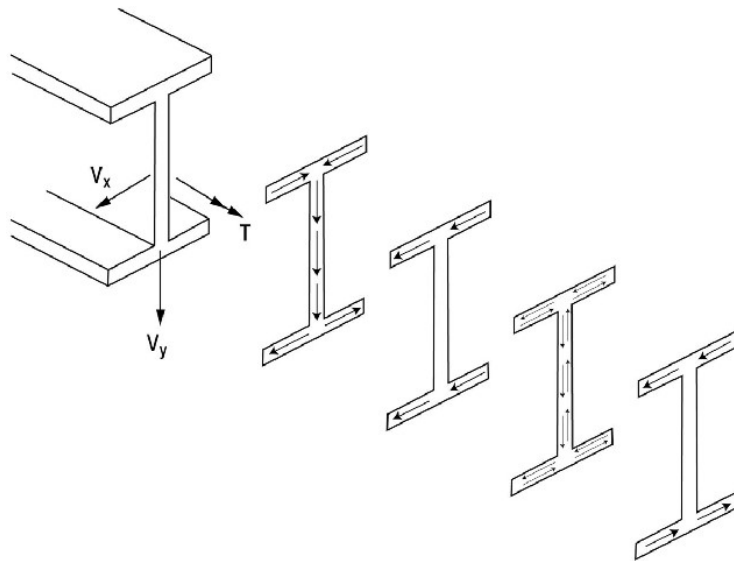
$$\text{Total Normal Stress} = \sigma = \frac{P}{A} + \frac{M_x y}{I_x} + \frac{M_y x}{I_y} + \text{Warping Normal Stress}$$

The bending moment (M) is obtained by computing the position of the elastic neutral axis, and integrating longitudinal stresses with respect to that axis.

Principal bending stresses ( $f_b$ ) are obtained by dividing the resulting bending moment by the appropriate section modulus.

Lateral bending stresses ( $f_l$ ) are obtained by subtracting the longitudinal stress at the flange center node from the stress at the flange tip.

## Step 2 – Resultant Shear Force



The resultant shear force ( $V$ ) is obtained by integrating the vertical shear stresses over the section.

$$\text{Total Shear Stress} = \tau = \frac{V_y Q_x}{I_x t} + \frac{V_x Q_y}{I_y t} + \text{St. Venant Torsion} + \text{Warping Torsion}$$

## Step 3 – Strength Design Checks – Positive Bending

Chapter 6.I0.6.2.2 – Composite Sections in Positive Flexure:

“Composite sections in □ horizontally curved steel-girder bridges shall be considered as noncompact sections and shall satisfy the requirements of Article 6.I0.7.2.”

Chapter 6.I0.7.2 – Noncompact Sections:

For the top flange (compression flange), verify Eq. 6.I0.7.2.1-1:  $f_{bu} \leq \phi_f F_{nc}$

For the bottom flange (tension flange), verify Eq. 6.I0.7.2.1-2:  $f_{bu} + \frac{1}{3} f_l \leq \phi_f F_{nt}$

Also, Chapter 6.I0.1.6 – Flange Stresses and Member Bending Moments:

For the bottom flange (discretely braced), verify Eq. 6.I0.1.6-1:  $f_l \leq 0.6 F_{yf}$

## Step 3 – Strength Design Checks – Negative Bending

Chapter 6.I0.6.2.3 – Composite Sections in Negative Flexure and Noncomposite Sections:

“Sections in  horizontally curved steel-girder bridges shall be considered as proportioned according to the provisions specified in Article 6.I0.8.”

Chapter 6.I0.8 – Flexural Resistance – Composite Sections in Negative Flexure and Noncomposite Sections:

For the bottom flange (discretely braced flange in compression), verify Eq. 6.I0.8.1-1:  $f_{bu} + \frac{1}{3}f_l \leq \phi_f F_{nc}$

Noting that this check has to be conducted for 2 limit states:

Flange local buckling (FLB)

Lateral-torsional buckling (LTB)

For the top flange (continuously braced flange in tension), verify Eq. 6.I0.8.1.3-1:  $f_{bu} \leq \phi_f R_h F_{yf}$

## Step 3 – Strength Design Checks – Bending

### Notes:

1. For bending checks, the hybrid factor ( $R_h$ ) is calculated automatically, following Chapter 6.10.1.10.1.
2. The web load-shedding factor ( $R_b$ ) is calculated automatically, following Chapter 6.10.1.10.2.
3. For lateral-torsional buckling (LTB) checks, the unbraced length ( $L_b$ ) is also calculated automatically, based on the cross-frame arrangement implemented in the mBrace3D model (X-frames and K-frames).  
A moment gradient factor ( $C_b$ ) of 1.0 is conservatively assumed.
4. Lateral bending stresses ( $f_l$ ) for the LL+IM load case are obtained from the finite element analysis (FEA) model directly, setting them as the response quantity for the influence analysis. This is a conservative approach, as the corresponding critical live load pattern may not be coincident with the one obtained for the composite bending moment.

## Step 3 – Strength Design Checks – Shear

### Chapter 6.I0.9.I – Shear Resistance:

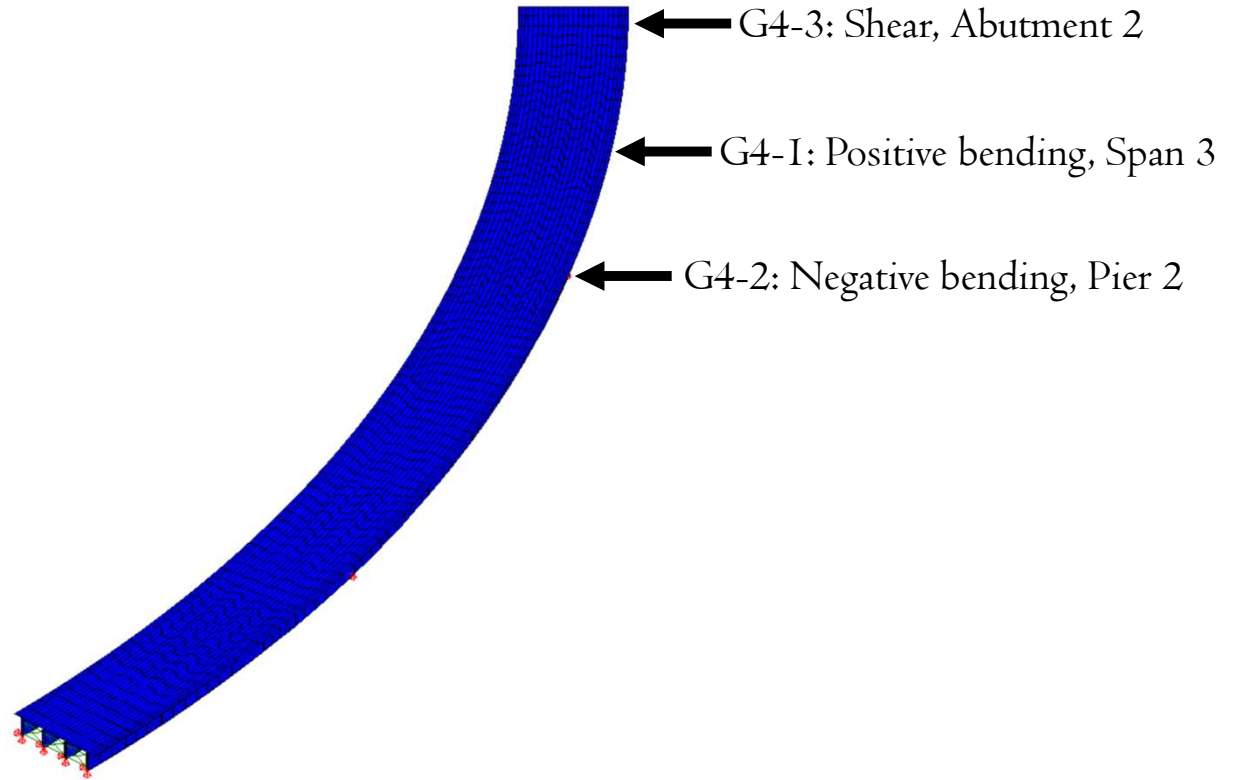
For the calculation of shear resistance, the transverse web stiffener spacing at the design location is determined using the connecting plates (for the X-frames and K-frames) included in the finite element model, together with additional user-defined transverse web stiffeners not explicitly represented in the mesh.

These supplementary stiffeners are used to limit the stiffener spacing at the design location, resulting in a more accurate assessment of panel shear resistance.

The nature of the panel (“end panel” or “interior panel”) is also automatically determined by the software.

Verify Eq. 6.I0.9.I-I:  $V_u \leq \phi_v V_n$

## AISC/NSBA Design Example 4 – Design Locations



Steel Bridge Design Handbook

DESIGN EXAMPLE 4

Three-Span Continuous Horizontally  
Curved Composite Steel  
I-Girder Bridge

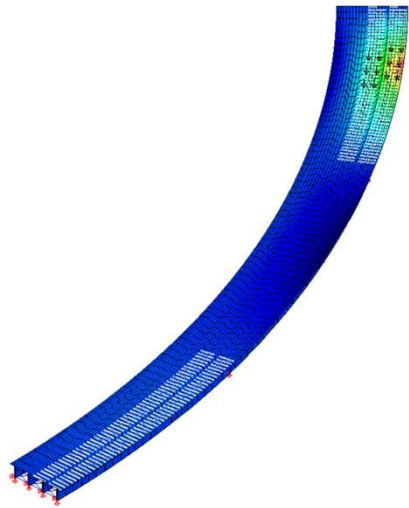
Compatible with the  
AASHTO LRFD Bridge Design Specifications, 10th Edition



Source: <https://www.aisc.org/bridges/bridge-resource-center/sbdh-example-4-three-span-continuous-horizontally-curved-composite-steel-i-girder-bridge/>

# AISC/NSBA Design Example 4 – Influence Surfaces

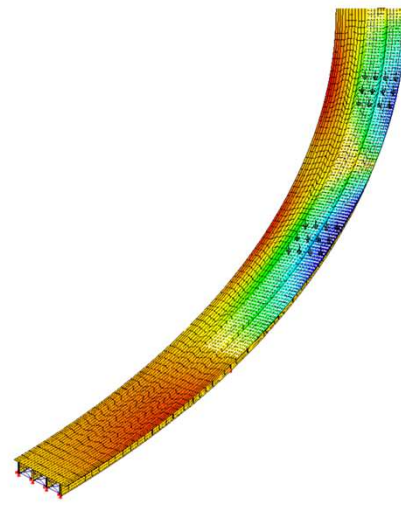
Influence surface  
Girder 4, y = 481, xs = 5, M  
VLO minimum value: 63839.75  
Max: 336.610  
Min: -89.778



Step: 1

Magnification factor: 0.0

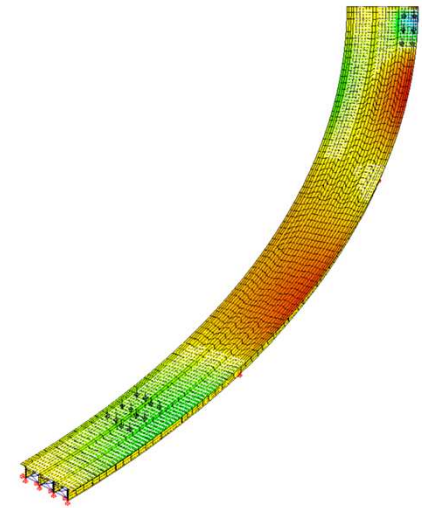
Influence surface  
Girder 4, y = 379, xs = 5, M  
VLO minimum value: -81315.00  
Max: 60.004  
Min: -233.886



Step: 1

Magnification factor: 0.0

Influence surface  
Girder 4, y = 537, xs = 5, V  
VLO minimum value: -75.15  
Max: 0.272  
Min: -0.866



Step: 1

Magnification factor: 0.0

G4-1: Positive bending, Span 3

$$M_{\max} = 5,487 \text{ kip} \cdot \text{ft}$$

(Published value: 5,125 kip · ft)

G4-2: Negative bending, Pier 2

$$M_{\min} = -6,776 \text{ kip} \cdot \text{ft}$$

(Published value: -6,726 kip · ft)

G4-3: Shear, Abutment 2

$$V_{\min} = -75.1 \text{ kips}$$

# AISC/NSBA Design Example 4 – Positive Bending

## I. Positive bending checks

### I.I Compute composite section properties

Composite section properties (positive bending)

Long-term (3n) section properties with transformed deck				Short-term (n) section properties with transformed deck			
Modular ratio	n	7.56		Modular ratio	n	7.56	
Equivalent reduced width (3n)	$b_{eff,3n}$	4.89	in	Equivalent reduced width (n)	$b_{eff,n}$	14.67	in
Equivalent reduced area (3n)	$A_{c,3n}$	44.02	in <sup>2</sup>	Equivalent reduced area (n)	$A_{c,n}$	132.07	in <sup>2</sup>
Location of the neutral axis	$NA_{3n}$	42.91	in	Location of the neutral axis	$NA_n$	28.26	in
Moment of inertia	$IA_{3n}$	213175	in <sup>4</sup>	Moment of inertia	$IA_n$	294105	in <sup>4</sup>
Distance from neutral axis to top of steel	$y_{tf,3n}$	30.91	in	Distance from neutral axis to top of steel	$y_{tf,n}$	16.26	in
Distance from neutral axis to bottom of steel	$y_{bf,3n}$	55.59	in	Distance from neutral axis to bottom of steel	$y_{bf,n}$	70.24	in
Distance from neutral axis to top of web	$D_{tf,3n}$	29.91	in	Distance from neutral axis to top of web	$D_{tf,n}$	15.26	in
Distance from neutral axis to bottom of web	$D_{bf,3n}$	28.41	in	Distance from neutral axis to bottom of web	$D_{bf,n}$	13.76	in
Top section modulus	$S_{tf,3n}$	6897	in <sup>3</sup>	Top section modulus	$S_{tf,n}$	18092	in <sup>3</sup>
Bottom section modulus	$S_{bf,3n}$	3835	in <sup>3</sup>	Bottom section modulus	$S_{bf,n}$	4187	in <sup>3</sup>

**Note:** Section properties slightly reduced from the published example, for demonstration purposes only.

## AISC/NSBA Design Example 4 – Positive Bending

- I. Positive bending checks
  - I.1 Compute composite section properties
  - I.2 Extract bending moments and lateral bending stresses from the FEA model

Load Factors		Moments							
					Top flange		Bottom flange		
$\gamma_{DC1-S}$	1.25	$M_{DC1-S}$	669.3	kip*ft	$f_{l_{DC1-S}}$	-0.96	ksi	0.46	ksi
$\gamma_{DC1-C}$	1.25	$M_{DC1-C}$	2656.9	kip*ft	$f_{l_{DC1-C}}$	-4.00	ksi	1.87	ksi
$\gamma_{DC2}$	1.25	$M_{DC2}$	547.2	kip*ft	$f_{l_{DC2}}$	0.77	ksi	0.47	ksi
$\gamma_{DW}$	1.5	$M_{DW}$	653.8	kip*ft	$f_{l_{DW}}$	-0.09	ksi	0.58	ksi
$\gamma_{LL+IM}$	1.75	$M_{LL+IM}$	5486.6	kip*ft	$f_{l_{LL+IM}}$	-0.44	ksi	3.26	ksi

**Note:** Section properties slightly reduced from the published example, for demonstration purposes only.

## AISC/NSBA Design Example 4 – Positive Bending

### I. Positive bending checks

I.1 Compute composite section properties

I.2 Extract bending moments and lateral bending stresses from the FEA model

I.3 Compute principal bending stresses

**Stresses**

		Top flange					Bottom flange				
		$f_{bu}$		$f_t$		$f_{bu}$		$f_t$		ksi	
		DC1-S	DC1-C	DC2	DW	LL+IM	$\Sigma$	DC1-S	DC1-C		DC2
Top flange	$f_{bu}$	DC1-S	-3.24	-4.05	ksi	Bottom flange	DC1-S	2.60	3.25	ksi	
		DC1-C	-12.87	-16.09	ksi		DC1-C	10.31	12.89	ksi	
		DC2	-0.95	-1.19	ksi		DC2	1.71	2.14	ksi	
		DW	-1.14	-1.71	ksi		DW	2.05	3.07	ksi	
		LL+IM	-3.64	-6.37	ksi		LL+IM	15.73	27.52	ksi	
		$\Sigma$	-21.84	-29.40	ksi		$\Sigma$	32.39	48.86	ksi	
	$f_t$	DC1-S	-0.96	-1.20	ksi		DC1-S	0.46	0.57	ksi	
		DC1-C	-4.00	-5.00	ksi		DC1-C	1.87	2.34	ksi	
		DC2	0.77	0.96	ksi		DC2	0.47	0.59	ksi	
		DW	-0.09	-0.13	ksi		DW	0.58	0.87	ksi	
		LL+IM	-0.44	-0.77	ksi		LL+IM	3.26	5.71	ksi	
		$\Sigma$	-4.71	6.13	ksi		$\Sigma$	6.65	10.09	ksi	

**Note:** Section properties slightly reduced from the published example, for demonstration purposes only.

## AISC/NSBA Design Example 4 – Positive Bending

- I. Positive bending checks
  - I.1 Compute composite section properties
  - I.2 Extract bending moments and lateral bending stresses from the FEA model
  - I.3 Compute principal bending stresses
  - I.4 Verify the strength design equations

Strength check, compression flange (Eq. 6.10.7.2.1-1)				Strength check, tension flange (Eq. 6.10.7.2.1-2)				Strength check, lateral bending (Eq. 6.10.1.6-1)			
	$f_{bu}$	-29.40	ksi	$f_{bu}$	48.86	ksi					
	$F_{nc}$	50	ksi	$f_l$	10.09	ksi			$f_l$	10.09	ksi
	$\phi_f$	1.0		$f_{bu}+1/3f_l$	52.22	ksi			$F_{yf}$	50	ksi
Art. 6.5.4.2	$\phi_f F_{nc}$	50	ksi	$\phi_f$	1.0				$0.6F_{yf}$	30	ksi
Eq. 6.10.7.2.1-1	$f_{bu} \leq \phi_f F_{nc}$	0.59	OK	Eq. 6.10.7.2.1-2	$f_{bu}+1/3f_l \leq \phi_f F_{nt}$	1.04	NG	Eq. 6.10.1.6-1	$f_l \leq 0.6F_{yf}$	0.34	OK

**Note:** Section properties slightly reduced from the published example, for demonstration purposes only.

## AISC/NSBA Design Example 4 – Negative Bending

### 2. Negative bending checks

#### 2.1 Compute composite section properties

#### Composite section properties (negative bending)

Long-term (3n) section properties with longitudinal steel reinforcement				Short-term (n) section properties with longitudinal steel reinforcement			
Location of the neutral axis	$NA_{3n}$	57.01	in	Location of the neutral axis	$NA_n$	56.18	in
Moment of inertia	$IA_{3n}$	318511	$in^4$	Moment of inertia	$IA_n$	327375	$in^4$
Top section modulus	$S_{tf,3n}$	6848	$in^3$	Top section modulus	$S_{tf,n}$	7167	$in^3$
Bottom section modulus	$S_{bf,3n}$	7409	$in^3$	Bottom section modulus	$S_{bf,n}$	7471	$in^3$

## AISC/NSBA Design Example 4 – Negative Bending

### 2. Negative bending checks

2.1 Compute composite section properties

2.2 Extract bending moments and lateral bending stresses from the FEA model

Load Factors		Moments			Lateral bending stresses			
					Top flange		Bottom flange	
$\gamma_{DC1-S}$	1.25	$M_{DC1-S}$	-1960.5	kip*ft	$f_{l_{DC1-S}}$	0.32 ksi	-0.45	ksi
$\gamma_{DC1-C}$	1.25	$M_{DC1-C}$	-6451.5	kip*ft	$f_{l_{DC1-C}}$	1.55 ksi	-1.44	ksi
$\gamma_{DC2}$	1.25	$M_{DC2}$	-1447.1	kip*ft	$f_{l_{DC2}}$	0.48 ksi	-0.19	ksi
$\gamma_{DW}$	1.5	$M_{DW}$	-1450.6	kip*ft	$f_{l_{DW}}$	0.12 ksi	-0.27	ksi
$\gamma_{LL+IM}$	1.75	$M_{LL+IM}$	-6776.3	kip*ft	$f_{l_{LL+IM}}$	0.67 ksi	-1.18	ksi

## AISC/NSBA Design Example 4 – Negative Bending

### 2. Negative bending checks

2.1 Compute composite section properties

2.2 Extract bending moments and lateral bending stresses from the FEA model

2.3 Compute principal bending stresses

**Stresses**

		Top flange					Bottom flange				
Top flange	$f_{bu}$	DC1-S	3.52	4.40	ksi	$f_{bu}$	DC1-S	-3.19	-3.99	ksi	
		DC1-C	11.57	14.47	ksi		DC1-C	-10.50	-13.12	ksi	
		DC2	2.54	3.17	ksi		DC2	-2.34	-2.93	ksi	
		DW	2.54	3.81	ksi		DW	-2.35	-3.52	ksi	
		LL+IM	11.35	19.86	ksi		LL+IM	-10.88	-19.05	ksi	
		$\Sigma$	31.52	45.70	ksi		$\Sigma$	-29.26	42.61	ksi	
	$f_l$	DC1-S	0.32	0.40	ksi	$f_l$	DC1-S	-0.45	-0.57	ksi	
		DC1-C	1.55	1.94	ksi		DC1-C	-1.44	-1.80	ksi	
		DC2	0.48	0.60	ksi		DC2	-0.19	-0.23	ksi	
		DW	0.12	0.18	ksi		DW	-0.27	-0.41	ksi	
		LL+IM	0.67	1.16	ksi		LL+IM	-1.18	-2.07	ksi	
		$\Sigma$	3.13	4.28	ksi		$\Sigma$	3.53	5.07	ksi	

# AISC/NSBA Design Example 4 – Negative Bending

## 2. Negative bending checks

- 2.1 Compute composite section properties
- 2.2 Extract bending moments and lateral bending stresses from the FEA model
- 2.3 Compute principal bending stresses
- 2.4 Verify the strength design equations

**Strength check, tension flange (Eq. 6.10.8.1.3-1)**

	$f_{bu}$	45.70	ksi
	$F_{yf}$	50	ksi
Art. 6.5.4.2	$\phi_f$	1.0	
	$\phi_f R_h F_{yf}$	50	ksi
Eq. 6.10.8.1.3-1	$f_{bu} \leq \phi_f R_h F_{yf}$	0.91	OK

**Strength check, compression flange (Eq. 6.10.8.1.1-1)**

	$f_{bu}$	42.61	ksi
	$f_{l1}$	5.07	ksi
Eq. 6.10.1.10.2-1 & Eq. 6.10.1.10.2-3	$R_b$	1.0	
Eq. 6.10.8.2.3b-3	$r_t$	7.41	in
Eq. 6.10.8.2.3a-4	$L_p$	16.37	ft
Unbraced length	$L_b$	20.50	ft
	$C_b$	1.00	
Eq. 6.10.1.6-2	$L_b \leq 1.1L_p \sqrt{C_b R_b / (f_{bu} / F_{yc})}$	NO	
Eq. 6.10.8.2.3b-2	$F_e$	260.0	ksi
Eq. 6.10.1.6-4	$f_l$	5.16	ksi
	AF	1.017	
Eq. 6.10.8.2.2-3	$\lambda_f$	4.50	
Eq. 6.10.8.2.2-4	$\lambda_{pf}$	9.15	
Art. 6.10.8.2.2	$F_{yf}$	35.0	ksi
Eq. 6.10.8.2.2-5	$\lambda_{rf}$	16.12	
Art. 6.10.8.2.2	$F_{nc,FLB}$	50.00	ksi
Eq. 6.10.8.2.3a-5	$L_r$	55.9	ft
Art. 6.10.8.2.3a	$F_{nc,LTB}$	48.43	ksi
	$f_{bu} + 1/3f_l$	44.33	ksi
Eq. 6.10.8.1.1-1, FLB	$f_{bu} + 1/3f_l \leq \phi_f F_{nc}$	0.887	OK
Eq. 6.10.8.1.1-1, LTB	$f_{bu} + 1/3f_l \leq \phi_f F_{nc}$	0.915	OK

## AISC/NSBA Design Example 4 – Shear

### 3. Shear check

3.1 Extract the shear forces from the FEA models

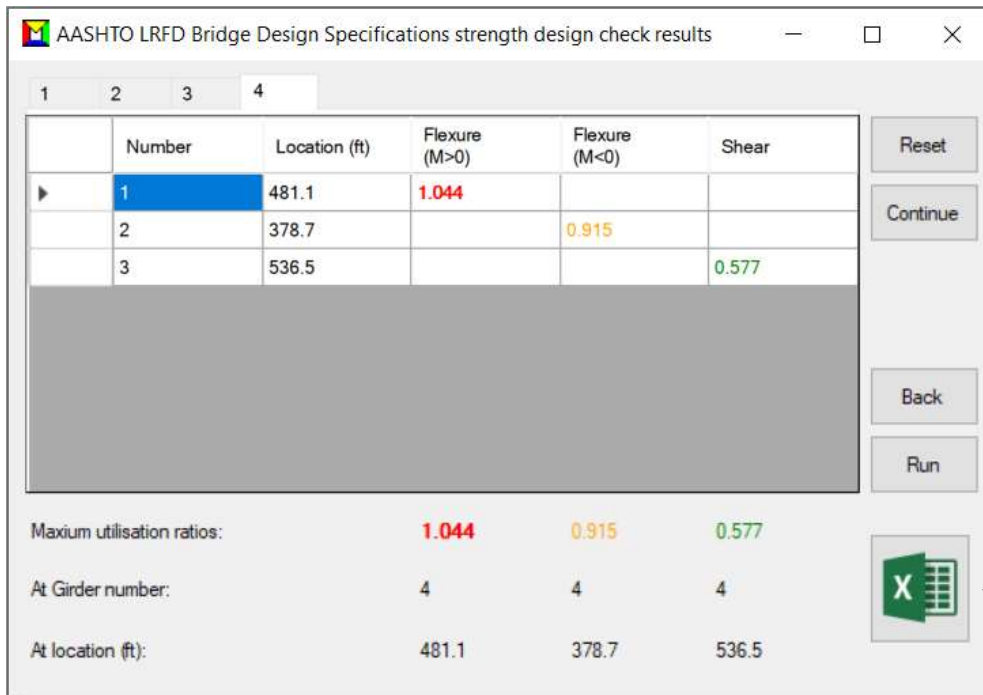
3.2 Calculate the shear resistance  
and verify the strength design equation

Applied shear		
$V_{DC1-S}$	-22.8	kips
$V_{DC1-C}$	-87.1	kips
$V_{DC2}$	-20.1	kips
$V_{DW}$	-19.2	kips
$V_{LL+IM}$	-75.1	kips
$V_u$	322.8	kips

Shear resistance (Eq. 6.10.9.1-1)			
	$F_{yw}$	50	ksi
	$E$	29000	ksi
	$D$	84	in
	$t_w$	0.5625	in
	$D/t_w$	149.3	
Transverse stiffener spacing	$d_o$	84	in
	Interior or end panel?	END	
Eq. 6.10.9.3.2-7	$k$	10.0	
Art. 6.10.9.3.2	$C$	0.408	
Eq. 6.10.9.3.3-2	$V_p$	1370.3	kips
Eq. 6.10.9.3.3-1	$V_n$	559.5	kips
Art. 6.5.4.2	$\phi_v$	1.0	
	$\phi_v V_n$	559.5	kips
Eq. 6.10.9.1-1, end panels	$V_u \leq \phi_v V_n$	0.58	OK
	$b_{fc}$	21	in
	$t_{fc}$	1	in
	$b_{ft}$	20	in
	$t_{ft}$	1	in
Eq. 6.10.9.3.2-1	$2Dt_w / (b_{fc}t_{fc} + b_{ft}t_{ft}) \leq 2.5$	YES	
Eq. 6.10.9.3.2-3	$V_p$	1370.25	kips
Eq. 6.10.9.3.2-2	$V_n$	1058.3	kips
	$\phi_v V_n$	1058.3	kips
Eq. 6.10.9.1-1, interior panels	$V_u \leq \phi_v V_n$	0.31	OK

## AISC/NSBA Design Example 4 – Summary

A summary of the design checks is provided in the software, together with an Excel button to open the corresponding design checks.



The screenshot shows the 'AASHTO LRFD Bridge Design Specifications strength design check results' window. It features a table with columns for 'Number', 'Location (ft)', 'Flexure (M>0)', 'Flexure (M<0)', and 'Shear'. The first row is highlighted in blue. Below the table, there are summary statistics for maximum utilisation ratios and girder numbers at specific locations. An Excel icon button is located at the bottom right of the window.

	Number	Location (ft)	Flexure (M>0)	Flexure (M<0)	Shear
▶	1	481.1	1.044		
	2	378.7		0.915	
	3	536.5			0.577

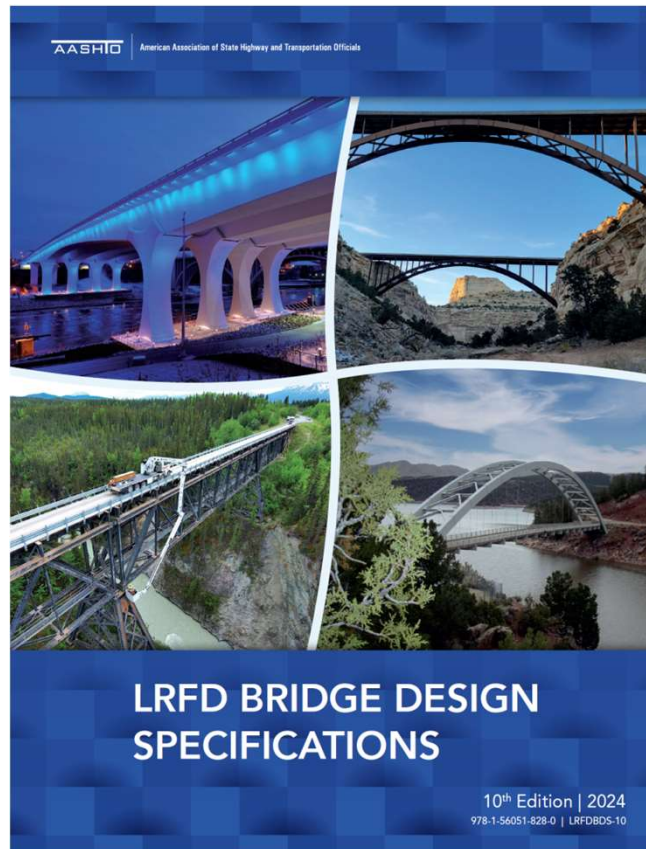
Maximum utilisation ratios:	1.044	0.915	0.577
At Girder number:	4	4	4
At location (ft):	481.1	378.7	536.5

Button to open the Excel spreadsheets showing the strength design checks at all design locations

## Concluding Remarks

1. The latest software release (v4.0) now implements the bending and shear strength design checks in compliance with the latest edition of the AASHTO LRFD Bridge Design Specifications (10<sup>th</sup> Edition, 2024).
2. As mBrace3D implements refined 3D shell models, this means optimal accuracy compared to more traditional beam or grid models.
3. Design equations are verified by extracting the results from different load cases: DCI-S, DCI-C, DC2, DW, and LL+IM.
4. Steel grades as well additional transverse web stiffeners and longitudinal deck reinforcement are defined prior to running the design checks.
5. 2<sup>nd</sup> order lateral bending stresses are evaluated using the AASHTO 6.10.1.4 Equation, where the unbraced length at the design location is calculated automatically based on the cross-frame arrangement specified by the user.
6. Additional parameters, such as the hybrid factor, web load-shedding factor, and shear panel type are also evaluated automatically.
7. The method implemented is the opposite of a black box approach, as all the calculations can be verified in Excel if necessary. These calculations may also be used in calculation reports for design documentation.

# Reference



American Association of State Highway and Transportation Officials (2024). "LRFD Bridge Design Specifications (10<sup>th</sup> ed.)". Washington, D.C.